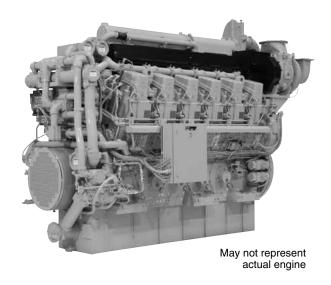
# **CATERPILLAR®**

C280-12 MARINE 3640 ekW AUXILIARY & DIESEL 3800 bkW (5096 bhp) ELECTRIC PROPULSION 60 Hz @ 900 rpm



### **SPECIFICATIONS**

### V-12, 4-Stroke-Cycle-Diesel

	MO II/EPA Tier 2 Compliant
	280 mm (11.0 in.)
Stroke	300 mm (11.8 in.)
Displacement	222 L (13,546 cu. in.)
Rotation (from flywheel e	end) Counterclockwise
Compression Ratio	13:1
Aspiration	Turbocharged-Aftercooled
Governor	Electronic
Low Idle Speed	350 rpm
Rated Speed	900 rpm
Oil Change Interval*	750 hr
Serial Number Prefix	TSJ
Cooling System	Keel or Heat Exchanger
Refill Capacities	
Cooling System	1400-1575 L (370-416 gal)
Lube Oil System	910 L (240 gal)

<sup>\*</sup>A new S•O•S<sup>sM</sup> analysis must be done to determine actual oil change intervals.

### STANDARD ENGINE EQUIPMENT

### Air Inlet System

Aftercooler — fresh water, corrosion resistant coated (air side); air inlet shutoff; breather — crankcase, top-mounted; turbocharger — engine oil lubricated

### **Control System**

Single ADEM<sup>™</sup> A3 electronic control unit (ECU) with electronic unit injector fuel system, rigid wiring harness (10 amp, 24 volt power required to drive ECUs)

### **Cooling System**

Engine coolant water drains

### **Exhaust System**

Dry, gas tight, exhaust manifold

### **Fuel System**

Distillate fuel (requires viscosity ranging from 1.4 cSt to 20 cSt at 38° C), fuel transfer pump (mounted on left-hand side), duplex fuel filters, electronically controlled unit injectors

### **Lube Oil System**

Centrifugal oil filters with single shutoff, service side engine mounted on cylinder block inspection covers includes installed oil lines and single shutoff valve, filters centrifuge bypass oil from the main lubricating oil pump, can be serviced with the engine running, oil filler and dipstick valve, oil pressure regulating valves, crankcase explosion relief valves

### Genera

Cat® yellow paint; gear-driven pumps: fuel, oil, jacket water, aftercooler/oil cooler water; service literature

Factory-designed systems built at Caterpillar ISO 9001:2000 certified facilities.

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## C280-12 MARINE AUXILIARY & **DIESEL ELECTRIC PROPULSION**

3640 ekW

### MARINE ENGINE PERFORMANCE

C280-12	DIESEL ENGINE 1	TECHNICA	AL DATA	CAT	<b>CATERPILLAR®</b>				
Genset	60 Hz		Prime	)					
			CERTIFICATION		IMO II/EPA N	MARINE TIE			
	900		TURBOCHARG	ER PART #:			157-5514		
	13:1 32		FUEL TYPE: RATED ALTITU	DE @ 25°C (	m):		Distillate 150		
	90		ASSUMED GEN			6)·	96		
	EUI		ASSUMED GEN				0.8		
EXHAUST MANIFOLD:	DRY		MEAN PISTON				9		
FIRING PRESSURE, MAXIMUM (kPa)	17300								
RATING		NOTES	LOAD	110%	100%	75%	50%		
ENGINE POWER		(2)	bkW	4180	3800	2850	1900		
GENERATOR POWER		(2)	ekW	4004	3640	2730	1820		
BMEP			kPa	2515	2286	1715	1143		
ENGINE EFFICIENCY	(ISO 3046/1)	(1)	%	43.1%	42.7%	41.1%	39.8%		
ENGINE EFFICIENCY	(NOMINAL)	(1)	%	41.8%	41.4%	39.8%	38.6%		
ENGINE DATA									
FUEL CONSUMPTION	(ISO 3046/1)	(1)	g/bkw-hr	195.8	198.0	206.2	212.8		
FUEL CONSUMPTION	(NOMINAL)	(1)	g/bkw-hr	199.6	201.9	210.2	217.0		
FUEL CONSUMPTION	(90% CONFIDENCE)	(1)	g/bkw-hr	201.7	204.1	212.7	219.8		
AIR FLOW (@ 25°C, 101.3 kPaa)			Nm3/min	374.7	363.6	320.6	232.7		
AIR MASS FLOW INLET MANIFOLD PRESSURE			kg/hr	25080 374.2	24336 356.2	21460 318.8	15572		
INLET MANIFOLD PRESSURE INLET MANIFOLD TEMPERATURE			kPa (abs) °C	374.2 41.7	356.2 42.5	39.0	233.7 36.4		
EXHAUST STACK TEMPERATURE			°C	383.4	374.4	365.8	371.4		
EXHAUST GAS FLOW (@ stack temp, 101.	3 kPa)		m3/min	779.9	745.7	650.9	492.7		
EXHAUST GAS MASS FLOW	5 tu u,		kg/hr	24612	24027	21363	15603		
EMISSIONS "NOT TO EXCEED	DATA"								
NOx (as NO) + THC (molecular weight of 13			g/bkW-hr	13.49	10.75	10.21	10.23		
NOx (as NO) + THE (molecular weight of 13	,		g/bkW-hr	12.65	9.93	9.41	9.10		
CO			g/bkW-hr	0.90	0.96	0.75	0.92		
THC (molecular weight of 13.018)			g/bkW-hr	0.84	0.82	0.80	1.13		
Particulates			g/bkW-hr	0.41	0.28	0.34	0.56		
EMISSIONS "NOMINAL DA	TA"								
Nox as NO2 + THC (molecular weight of 13.			g/bkW-hr	11.65	9.27	8.80	8.78		
Nox as NO2			g/bkW-hr	11.00	8.64	8.18	7.91		
СО			g/bkW-hr	0.69	0.74	0.58	0.71		
THC (molecular weight of 13.018)			g/bkW-hr	0.65	0.63	0.62	0.87		
Particulates			g/bkW-hr	0.29	0.20	0.24	0.40		
ENERGY BALANCE DAT	Δ 1								
FUEL INPUT ENERGY (LHV)	(NOMINAL)	(1)	KW	9995	9180	7157	4919		
HEAT REJ. TO JACKET WATER	(NOMINAL)	(3)	KW	823	774	651	514		
HEAT REJ. TO ATMOSPHERE	(NOMINAL)	(4)	KW	200	184	143	98		
HEAT REJ. TO OIL COOLER	(NOMINAL)	(5)	KW	406	384	332	280		
HEAT REJ. TO EXH. (LHV to 25°C)	(NOMINAL)	(3)	KW	2983	2761	2096	1457		
HEAT REJ. TO EXH. (LHV to 177°C)	(NOMINAL)	(3)	KW	2197	2126	1687	1139		
HEAT REJ. TO AFTERCOOLER	(NOMINAL)	(6) (7)	KW	1371	1251	1069	662		

CONDITIONS AND DEFINITIONS

ENGINE RATING OBTAINED AND PRESENTED IN ACCORDANCE WITH ISO 3046/1 AND SAE J1995 JAN90 STANDARD REFERENCE CONDITIONS OF 25°C, 100 KPA, 30% RELATIVE HUMIDITY AND 150M ALTITUDE AT THE STATED AFTERCOOLER WATER TEMPERATURE CONSULT ALTITUDE CURVES FOR APPLICATIONS ABOVE MAXIMUM RATED ALTITUDE AND/OR TEMPERATURE PERFORMANCE AND FUEL CONSUMPTION ARE BASED ON 35 API, 16°C FUEL HAVING A LOWER HEATING VALUE OF 42.780 KJ/KG USED AT 29°C WITH A DENSITY OF 838.9 G/LITER

### **NOTES**

- 1) FUEL CONSUMPTION TOLERANCE. ISO 3046/1 IS 0,  $\pm$  5% OF FULL LOAD DATA. NOMINAL IS  $\pm$  3 % OF FULL LOAD DATA 2) ENGINE POWER TOLERANCE IS  $\pm$  3 % OF FULL LOAD DATA.
- 2) ENGINE FOWER TOLERANCE IS ±3.% OF FULL LOAD DATA.

  3) HEAT REJECTION TO JACKET AND EXHAUST TOLERANCE IS ±10% OF FULL LOAD DATA. (heat rate based on treated water 4) HEAT REJECTION TO ATMOSPHERE TOLERANCE IS ±50% OF FULL LOAD DATA. (heat rate based on treated water 5) HEAT REJECTION TO LUBE OIL TOLERANCE IS ±20% OF FULL LOAD DATA. (heat rate based on treated water 6) HEAT REJECTION TO AFTERCOOLER TOLERANCE IS ±5% OF FULL LOAD DATA. (heat rate based on treated water

- 7) TOTAL AFTERCOOLER HEAT = AFTERCOOLER HEAT x ACHRF (heat rate based on treated water) 8) FUEL CONSUMPTION DATA IS WITHOUT SEA WATER PUMP.

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## C280-12 MARINE AUXILIARY & DIESEL ELECTRIC PROPULSION

3640 ekW

### MARINE ENGINE PERFORMANCE

C280-12

### **DIESEL ENGINE TECHNICAL DATA**



	50	0.94	0.91	0.88	0.86	0.83	0.81	0.78	0.76	0.74	0.71	0.69	0.67	0.65
	45	0.95	0.93	0.90	0.87	0.85	0.82	0.80	0.77	0.75	0.73	0.70	0.68	0.66
AIR	40	0.97	0.94	0.91	0.89	0.86	0.83	0.81	0.78	0.76	0.74	0.71	0.69	0.67
TO	35	0.98	0.96	0.93	0.90	0.87	0.85	0.82	0.80	0.77	0.75	0.73	0.70	0.68
TURBO	30	1.00	0.97	0.94	0.92	0.89	0.86	0.84	0.81	0.79	0.76	0.74	0.71	0.69
	25	1.00	0.99	0.96	0.93	0.90	0.88	0.85	0.82	0.80	0.77	0.75	0.73	0.70
(°C)	20	1.00	1.00	0.98	0.95	0.92	0.89	0.86	0.84	0.81	0.79	0.76	0.74	0.72
	15	1.00	1.00	0.99	0.96	0.93	0.91	0.88	0.85	0.83	0.80	0.78	0.75	0.73
	10	1.00	1.00	1.00	0.98	0.95	0.92	0.89	0.87	0.84	0.82	0.79	0.77	0.74
		0	250	500	750	1000	1250	1500	1750	2000	2250	2500	2750	3000
						ALTITUE	DE (METI	ERS ABO	OVE SEA	LEVEL)				

	AFTER	COOLER	HEAT RE	JECTION I										
	50	1.23	1.27	1.30	1.34	1.38	1.42	1.45	1.49	1.53	1.56	1.60	1.64	1.67
	45	1.18	1.22	1.25	1.29	1.32	1.36	1.39	1.43	1.46	1.50	1.53	1.57	1.61
AIR	40	1.13	1.17	1.20	1.23	1.27	1.30	1.34	1.37	1.40	1.44	1.47	1.50	1.54
TO	35	1.08	1.12	1.15	1.18	1.21	1.24	1.28	1.31	1.34	1.37	1.41	1.44	1.47
TURBO	30	1.03	1.06	1.10	1.13	1.16	1.19	1.22	1.25	1.28	1.31	1.34	1.37	1.40
	25	1.00	1.01	1.04	1.07	1.10	1.13	1.16	1.19	1.22	1.25	1.28	1.31	1.34
	20	1.00	1.00	1.00	1.02	1.05	1.07	1.10	1.13	1.16	1.19	1.21	1.24	1.27
	15	1.00	1.00	1.00	1.00	1.00	1.02	1.04	1.07	1.10	1.12	1.15	1.18	1.20
	10	1.00	1.00	1.00	1.00	1.00	1.00	1.00	1.01	1.04	1.06	1.09	1.11	1.14
						ALTITUE	DE (METI	ERS ABO	OVE SEA	LEVEL)				

### FREE\_FIELD MECHANICAL NOISE

### SOUND PRESSURE LEVEL dB(A)

DISTANCE FROM 15M 7M THE ENGINE 1M (M)

96 94.7 90.7 90.2 90.8 89.8 87.8 86.5 8	90.0									
	95.6	98	99.3	101.3	102.3	101.7	102.2	96.2	107	
00 70:2 00:2 01:7 00:0 01:0 02:0 01 7	84.1	86.5	87.8	89.8	90.8	90.2	90.7	94.7	96	
1 90 792 852 847 853 843 823 81 7	78.6	81	82.3	84.3	85.3	84.7	85.2	79.2	90	ı

Octave Band (Hz)

### FREE\_FIELD EXHAUST NOISE

### SOUND PRESSURE dB(A)

DISTANCE FROM THE ENGINE

15M	98	108.6	105.7	97.4	92.1	87.7	88.2	86.3	80.9
7M	104	116.4	113.0	103.7	98.9	94.0	95.0	93.6	88.2
1.5M	118	128.9	127.5	117.3	112.5	108.1	109.5	107.1	101.3
	Overall	63	125	250	500	1000	2000	4000	8000

Octave Band (Hz)

### **TOTAL DERATION FACTORS:**

This table shows the deration required for various air inlet temperatures and altitudes. Use this information to help determine actual engine power for your site. The total deration factor includes deration due to altitude and ambient temperature, and air inlet manifold temperature deration.

### AFTERCOOLER HEAT REJECTION FACTORS:

Aftercooler heat rejection is given for standard conditions of 25°C and 150 m altitude. To maintain a constant air inlet manifold temperature, as the air to turbo temperature goes up, so must the heat rejection. As altitude increases, the turbocharger must work harder to overcome the lower atmospheric pressure. This increases the amount of heat that must be removed from the inlet air by the aftercooler. Use the aftercooler heat rejection factor to adjust for ambient and altitude conditions. Multiply this factor by the standard aftercooler heat rejection.

### **GENERATOR EFFICIENCY:**

Generator power determined with an assumed generator effeciency of 96% [generator power = engine power x 0.96]. If the actual generator efficiency is less than 96% [and greater than 94.5%], the generator power [ekW] listed in the technical data can still be achieved. The BSFC values must be increased by a factor.

The factor is a percentage = 96% - actual generator efficiency.

Data determined by methods similar to ISO Standard DIS-8528-10. Accuracy Grade 3.

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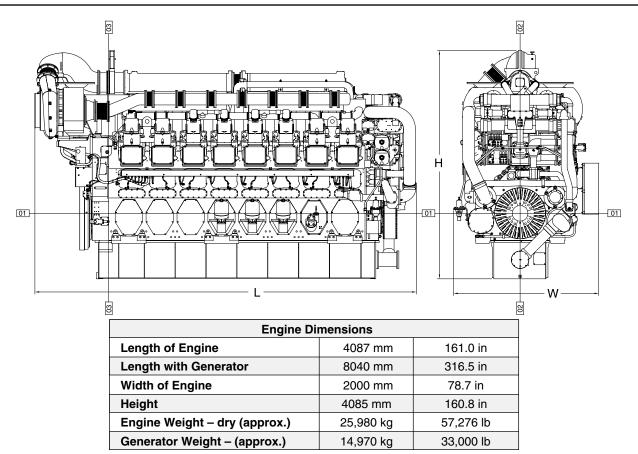
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# C280-12 MARINE AUXILIARY & DIESEL ELECTRIC PROPULSION

3640 ekW

### **DIMENSIONS**



### RATING DEFINITIONS AND CONDITIONS

**Continuous Power** — 8,000 hrs./yr., for applications with load factors less than or equal to 100%. 10% overload available.

Ratings are based on SAE J1995/ISO3046 standard conditions of 100 kPa (29.61 in. Hg), 25°C (77°F), and 30% relative humidity at the stated charge air cooler water temperature. Ratings also meet classification society maximum temperature requirements of 45°C (113°F) air temperature to the turbocharger and 32°C (90°F) seawater temperature without derate.

Additional ratings may be available for specific customer requirements. Consult your Cat representative for additional information.

**Fuel rates** are based on 35° API, 16°C (60°F) fuel used at 29°C (85°F) with a density of 838.9 g/liter (7.001 lbs/U.S. gal). Lower Heat Value (LHV) of 42 780 kJ/kg (18,390 Btu/lb). Tolerance is +5%. Includes all engine mounted pumps. BSFC without pumps is 3% less.

Marine Certification — Ratings are marine classification society approved by ABS, BV, CCS, DnV, GL, LRS, and RINA. These societies have also granted C280 factory line production approval which eliminates requirement for society surveyor witness test.

Performance data is calculated in accordance with tolerances and conditions stated in this specification sheet and is only intended for purposes of comparison with other manufacturers' engines. Actual engine performance may vary according to the particular application of the engine and operating conditions beyond Caterpillar's control.

Power produced at the flywheel will be within standard tolerances up to 49°C (120°F) combustion air temperature measured at the air cleaner inlet, and fuel temperature up to 52°C (125°F) measured at the fuel filter base. Power rated in accordance with NMMA procedure as crankshaft power. Reduce crankshaft power by 3% for propeller shaft power.

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